

Actuator and Sensor Placement

Sharon Padula

ASCAC Methods Development

Peer Review

November 27-29, 2001

Morphing Project (1998- 2002)

Long-Term Vision: Aerospace Vehicles that Efficiently Adapt to Handle Diverse Mission Scenarios

NASA Morphing Project Objective:

Develop and assess innovative, advanced technologies and integrated concepts to enable efficient, multi-point adaptability in air and space vehicles

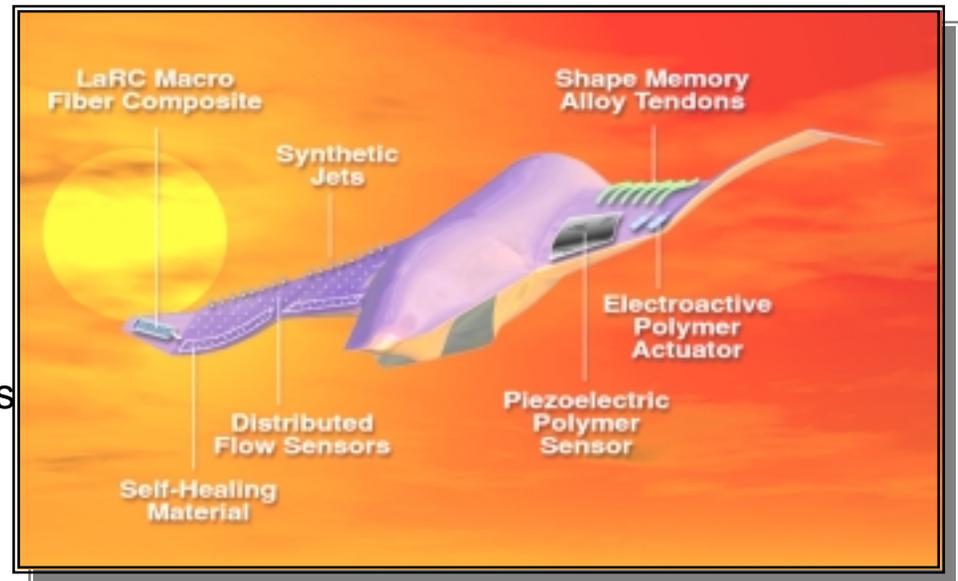
Morphing

Technical Challenges:

- Active materials for sensors and actuators
- Compliant, load-bearing structures
- Unsteady aerodynamics
- Non-intrusive, efficient electronics
- Optimization and Controls
- Manufacturability and Reliability
- New structural and vehicle control concepts

Technical Approach:

- Create Innovative Technologies
- Address Application Issues
- Demonstrate Performance
- Devise Revolutionary Concepts



Actuator and Sensor Placement Survey Paper Completed in 1999

- **NASA TM by Padula and Kincaid**
- **Langley Technical Report Server**
 - <http://techreports.larc.nasa.gov/ltrs/ltrs.html>
- **Includes more than 50 references**
- **Wide variety of applications**
 - Aerospace
 - Nuclear power
 - Water resources
 - Cancer Treatment
 - Etc.
- **Very few use formal optimization techniques**

Outline

- **Optimization Methods for Selecting Locations**
- **Sample Applications**
 - Vibration suppression and aeroelastic control
 - Interior noise control
 - Novel 3-axis flight control system
- **Results**

Collaborations

Years	Application	Branch	Contact
1990-1992	Vibration Suppression	Structural Dynamics / SMC	Chris Sandridge
1995-1997	Interior Noise	Structural Acoustics/ AAAC	Dan Palumbo
1998-2000	Flight Control	Dynamics & Control / AirSC	Dave Raney
2000-2001	Adaptive Aeroelastic Demonstrator	Aeroelasticity / SMC	Rob Scott

Choosing Actuator and/or Sensor Locations

- **Pick a large number (N) of possible locations**
- **Choose subset of M locations such that $M \ll N$**
- **Estimate the effectiveness of one subset compared to another**
- **Use a heuristic search method to hunt for the best subsets**

Heuristic Search Methods

Global search methods

Enumeration - Useful if the number of combinations is small

Local search methods

Random Trials - Easy to code but may reevaluate same subset

Simulated Annealing - Easy to code but converges slowly

Tabu Search - Easy to include knowledge of the problem

Genetic Algorithms - Ideal for parallel processing
- Finds widely scattered solutions

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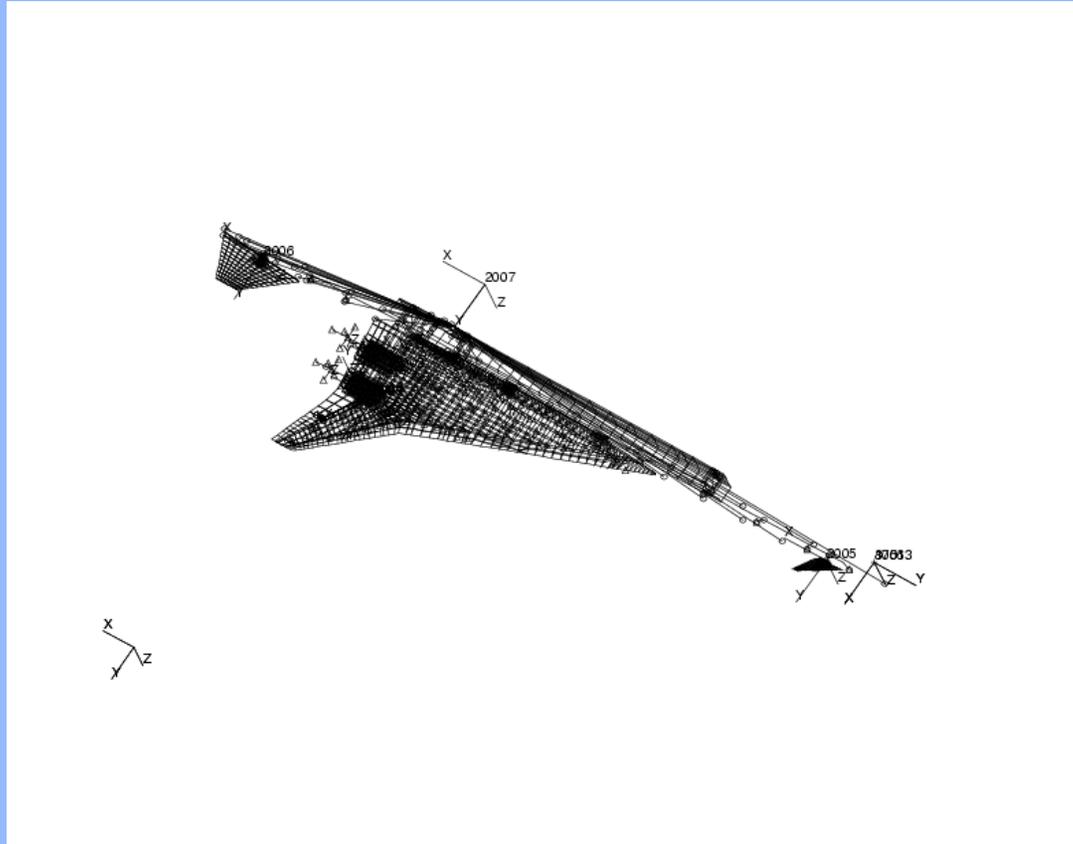
Initial Application Vibration Suppression for Large Space Structures



- **Actuators:** smart struts
- **Sensors:** smart struts
- **Challenge:** control all important vibration modes
- **Goal:** maximize the minimum damping ratio over all modes
- **Note:** Similar methods can be used for aeroelastic tailoring of structures

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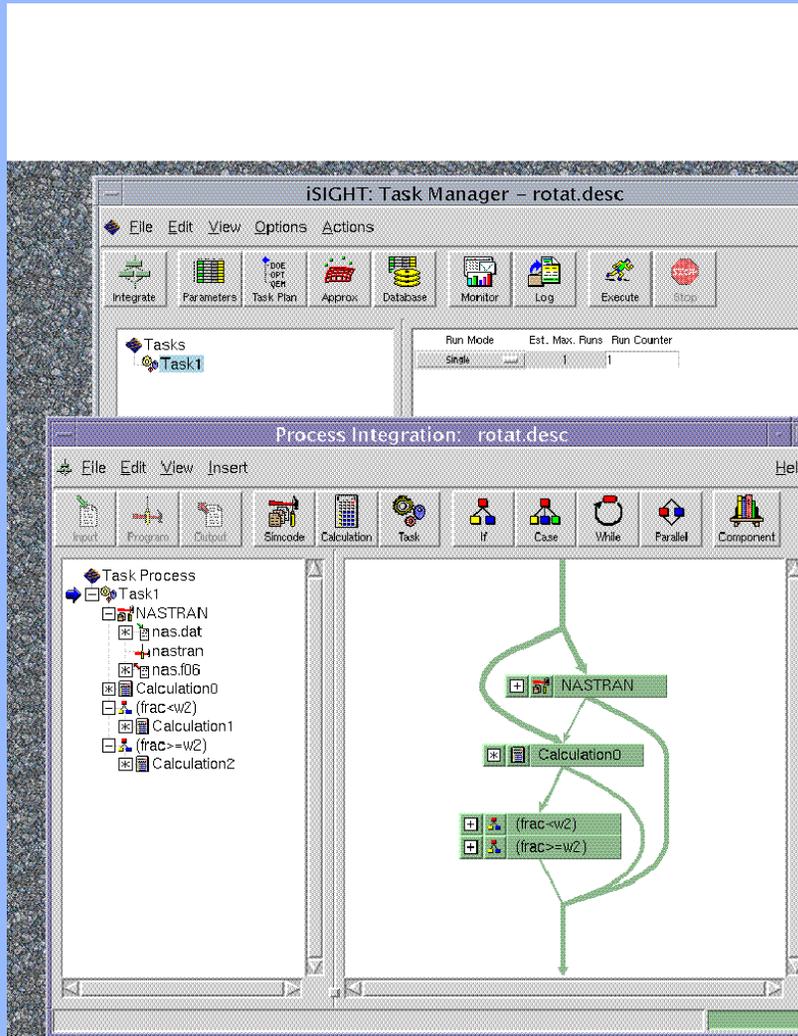
Develop procedures to optimize composite ply lay-up and actuator locations for enhanced aeroelastic performance.



NASTRAN finite element model of Langley Adaptive Aeroelastic Demonstrator (LAAD)

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iSIGHT Tools for Vibration Suppression

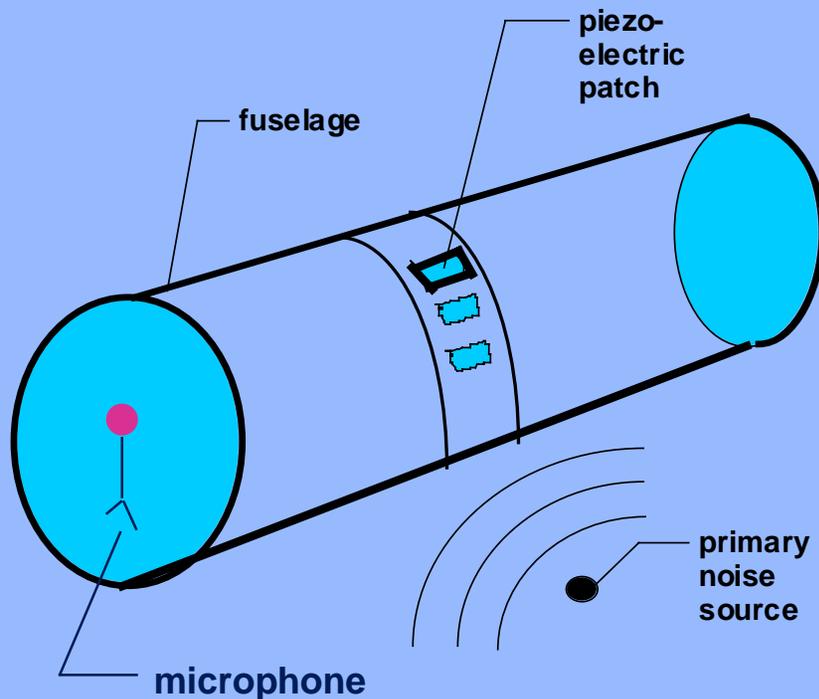


- Original system hand-coded UNIX scripts and C preprocessors
- iSIGHT framework is better:
 - Parse inputs and outputs
 - Choose Optimizer (e.g. Simulated Annealing and Genetic Algorithms)
 - Change objectives and constraints easily
 - Save results in database
 - Monitor progress
 - Parallel processing

Outline

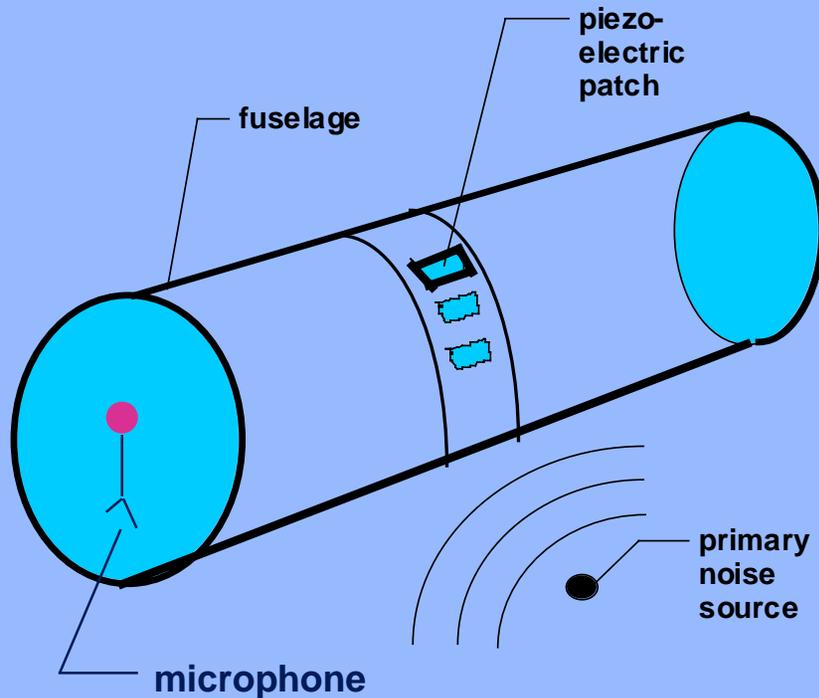
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Interior Noise Control



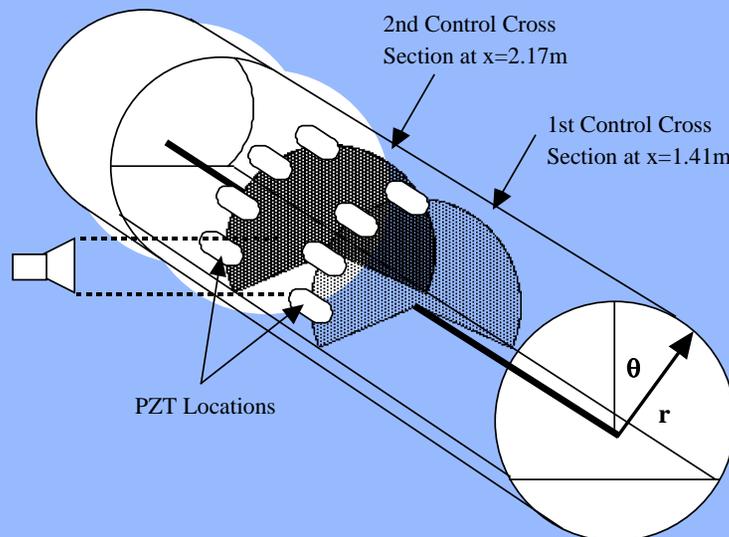
- **Actuators:** piezoelectric patches bonded to fuselage
- **Sensors:** microphones inside aircraft cabin
- **Challenge:** best locations for actuators and sensors
- **Goal: global noise reduction**

Predicting Noise Reduction



- Measure noise at each mic with no actuators
- Measure noise again using one actuator at a time
- Assume linear superposition
- Use measured data to estimate noise reduction for subsets of actuators and sensors
- Use tabu search to find best subsets

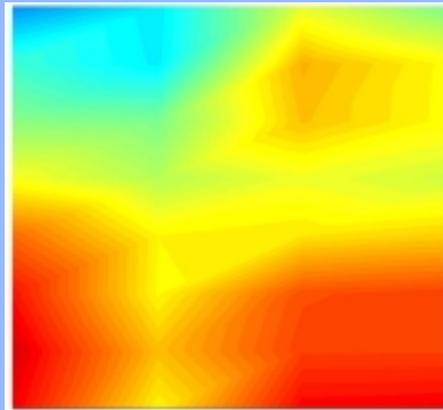
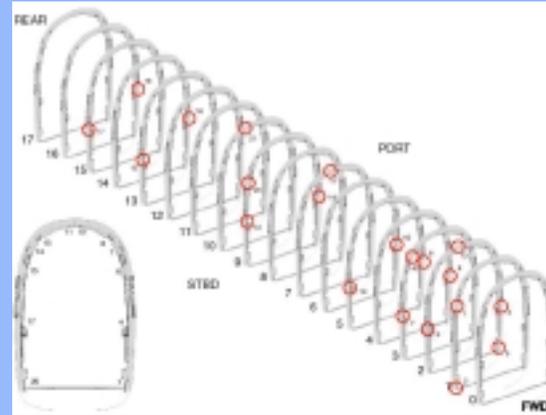
Interior Noise Control - Laboratory Tests



- Actuators: Pick 4 out of 8
- Sensors: Pick 8 out of 462
- **Goal: Reduce noise at 462 microphone locations**
- Tests show that tabu search picks better locations than manual method

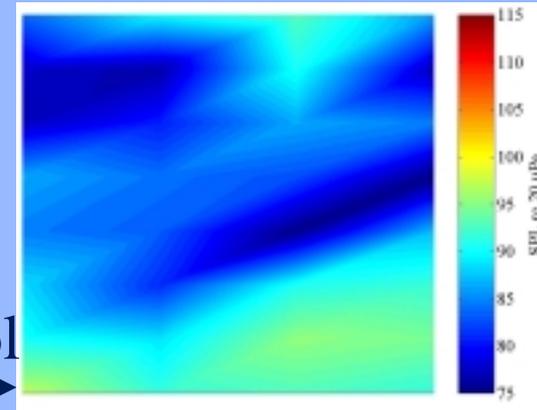
Interior Noise Control - Flight Tests

Selected 21 out of 84 possible actuators



No control

With control



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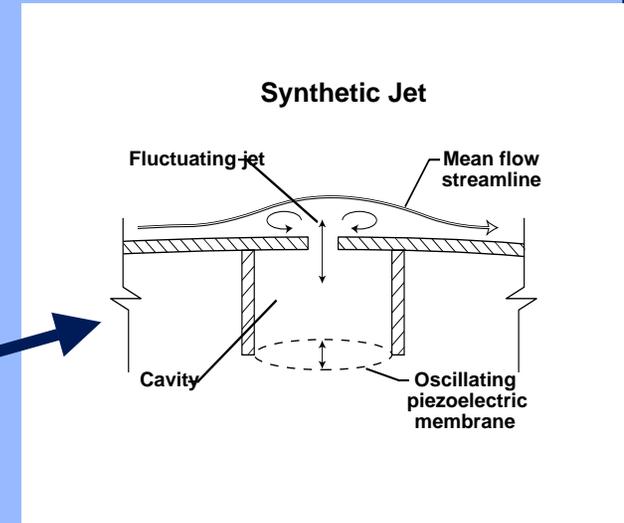
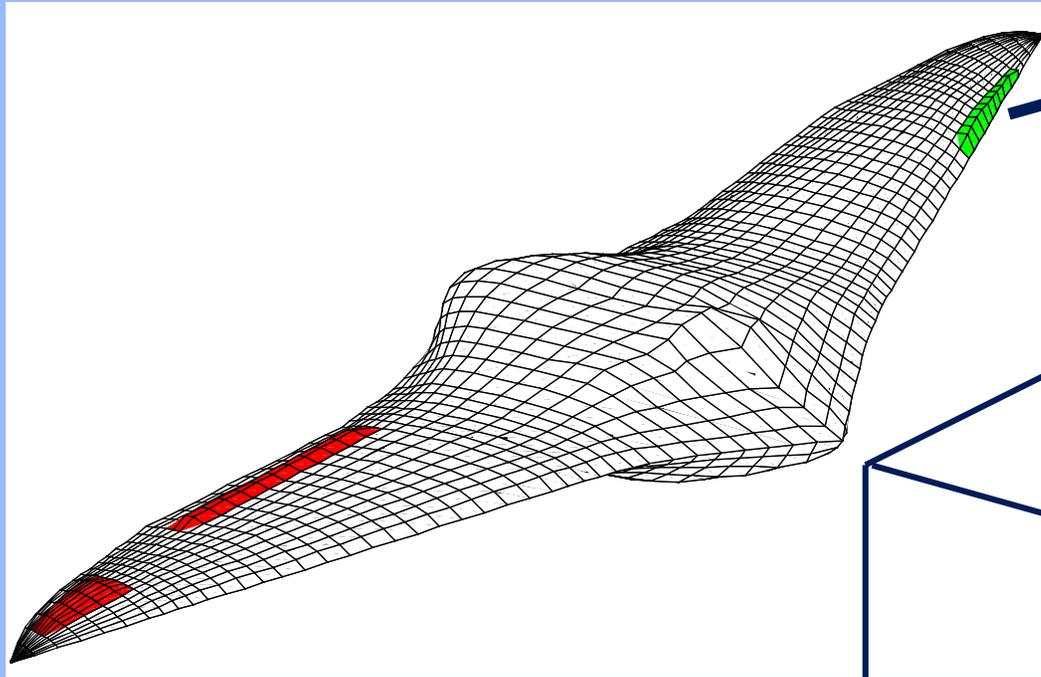
Novel Three-axis Flight Controller



- **Actuators:** zero mass-flow jets which change apparent shape of wing
- **Sensors:** N/A
- **Challenge:** provide uncoupled pitch, roll and yaw with fewest number of actuators
- **Goal:** stealth and mild maneuvering

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Objective: Minimize number of effectors to provide uncoupled pitch, roll, and yaw.



pitch

roll

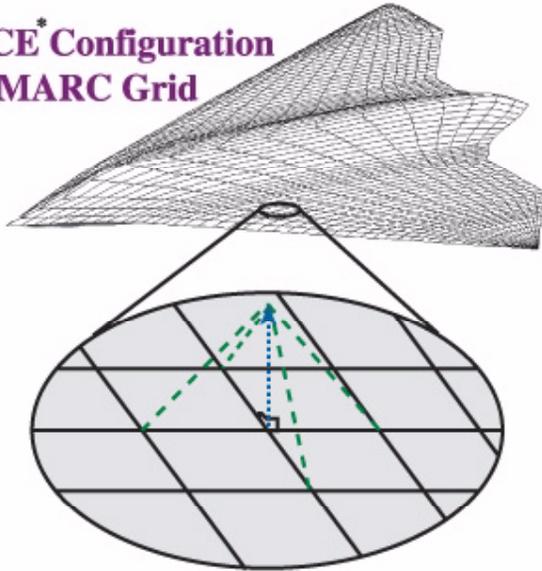
yaw

Optimized Effector Locations

- Initial demonstration problem has 34 potential locations - implies 1.7 billion combinations
- Exhaustive search would require 286 million hours of computer time
- MDO technology reduced actual computer time to 1 hour
- Technology used
 - Automatic differentiation
 - Approximate analysis
 - Genetic algorithm

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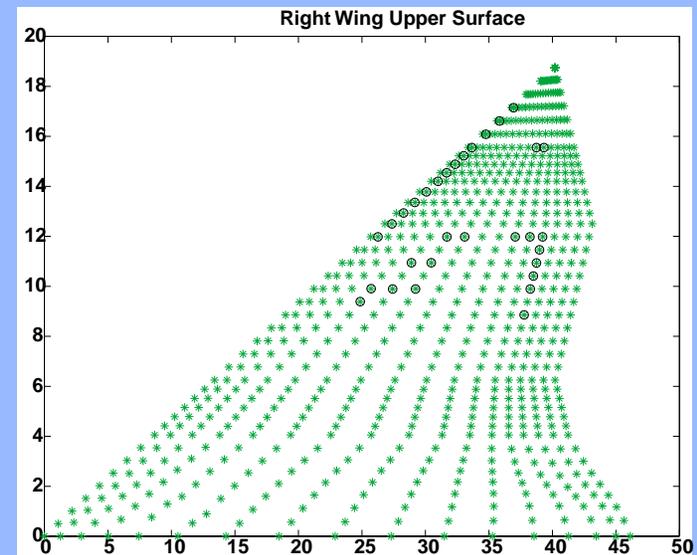
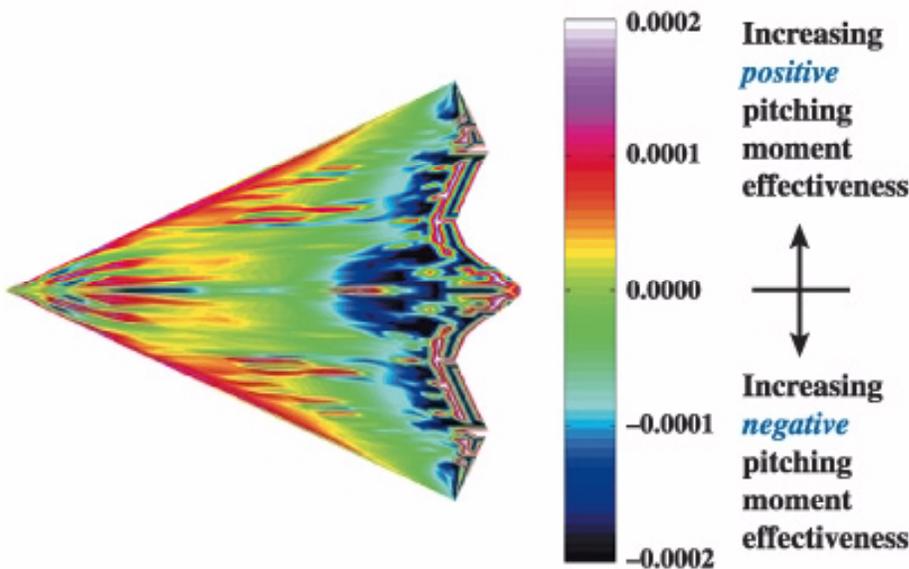
ICE* Configuration
PMARC Grid



1.) Control Effectiveness Determined by Automatic Differentiation of PMARC - MDOB

2.) Candidate Effector Locations Selected by Controls Expert - DCB

3.) Effector Locations Optimized using Genetic Algorithm and MATLAB - MDOB



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Assess Control Effectiveness

- **Use MATLAB to evaluate each effector array**
- **Place arrays on upper and/or lower surface of semi-span model (i.e., assume symmetric pairs of arrays)**
- **Estimate moments for Roll (C_l), Pitch (C_m) and Yaw (C_n)**
- **Change height of effectors to produce desired uncoupled moment (e.g. C_l without C_m or C_n)**

GA Selects Best Effector Suite from Combinations of Potential Arrays

Objective function = No. of effectors + penalty

Roll Penalty={0 or 150}

Yaw Penalty={0 or 150}

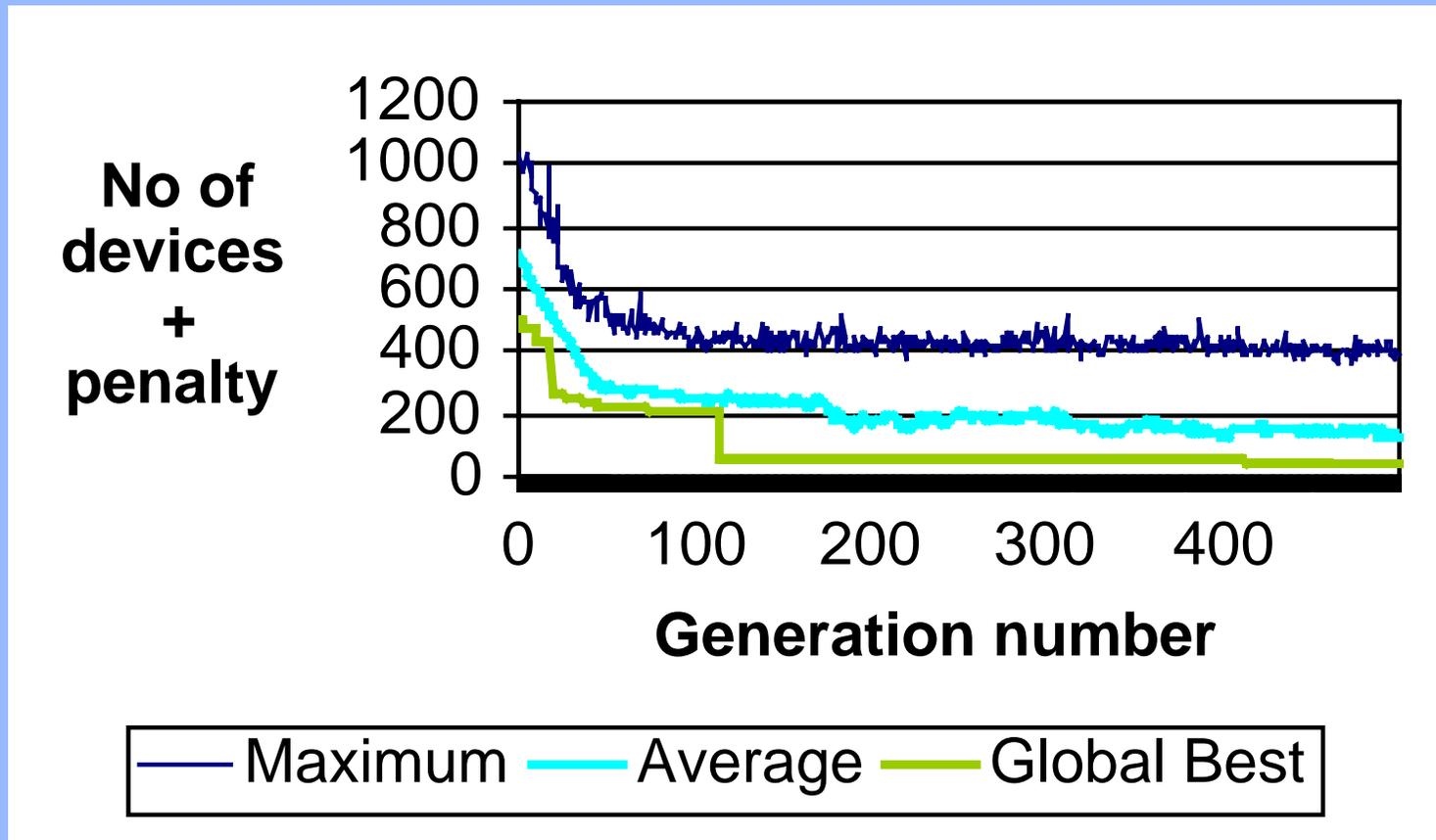
Pitch Penalty={0 or 150}

Evaluate for
roll > .0006

Evaluate for
pitch > .0005

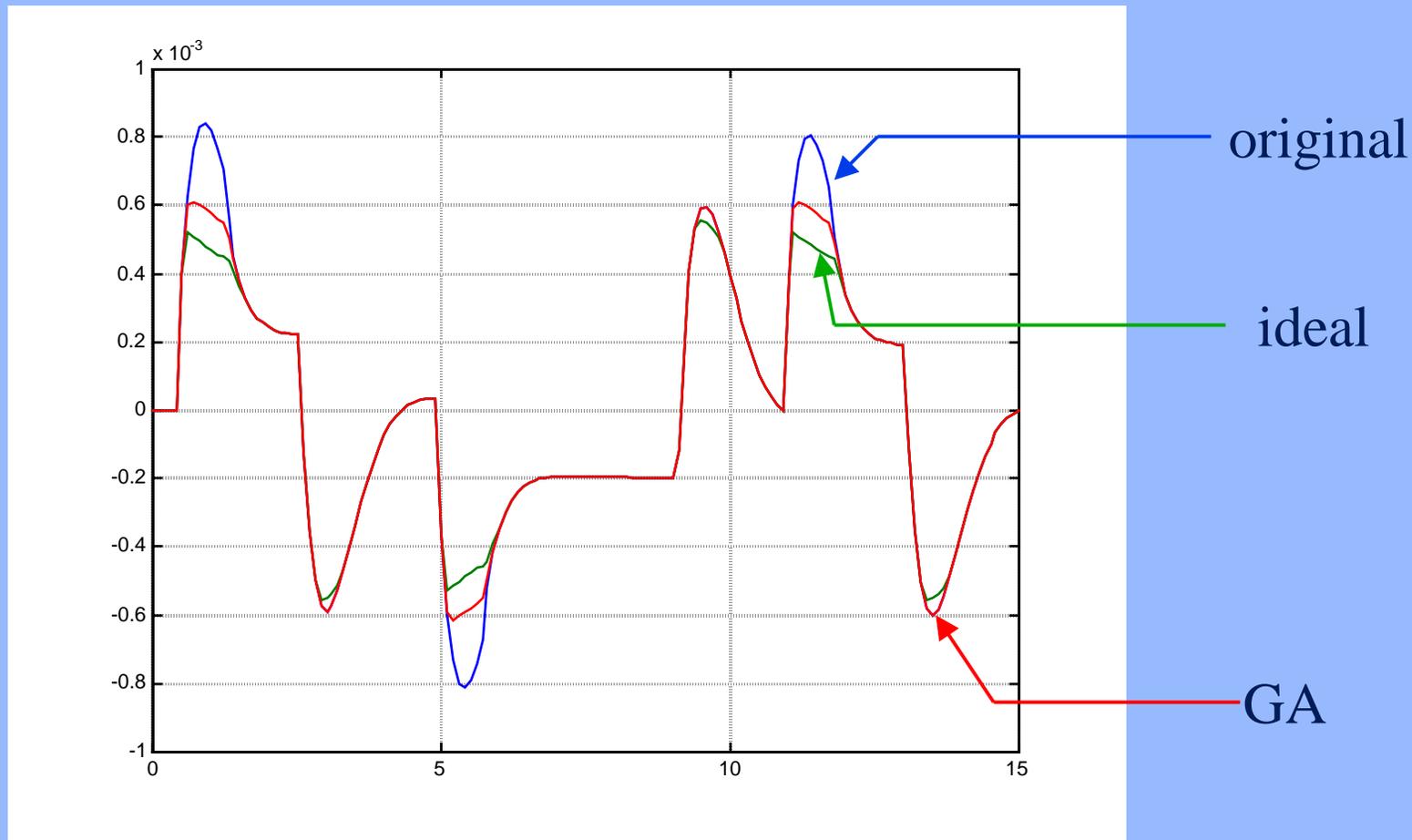
Evaluate for
yaw > .0003

Typical GA Convergence



Test of Simulated Controller

Effectors chosen by GA perform slightly better than original



Note: Simulated +/- 20 deg. bank angle doublet with roll rate < 10 deg/s

Summary

- **Combinatorial optimization finds best actuator and sensor locations in wide set of applications**
- **Application to interior noise control is most mature. Successfully used and flight tested by Structural Acoustics Branch**
- **Application to 3-axis Flight Control problems delivered to Dynamics and Control Branch**
- **Application to vibration damping and gust load alleviation explored - future funding uncertain**

Sharon Padula Actuator Placement - Selected References

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